

NARSIMHA REDDY ENGINEERING COLLEGE UGC AUTONOMOUS INSTITUTION

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Department of Mechanical Engineering

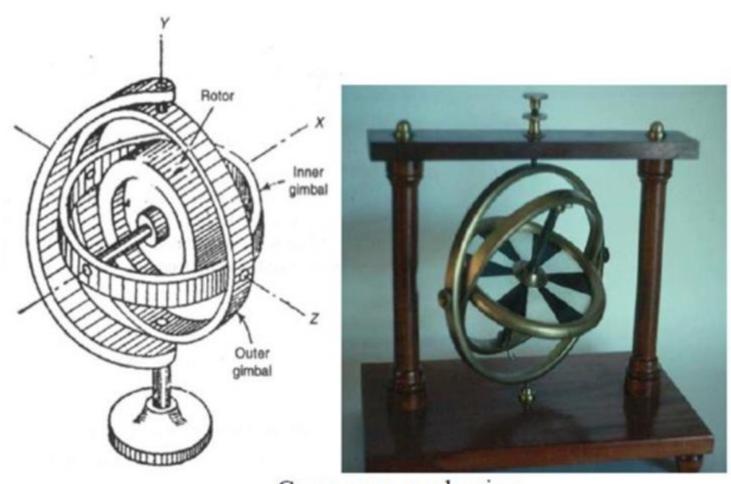
Mr. R Sai Syam, Asst.Professor,

Subject: Dynamics of Machinery

Code: 23ME501

Gyroscope

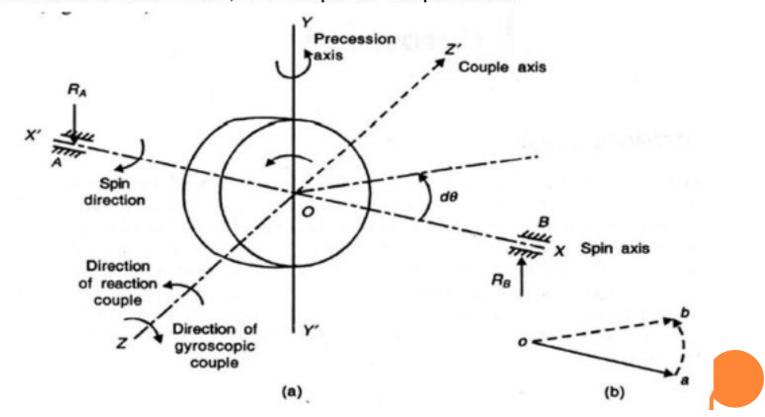
A gyroscope consists of a rotor mounted in the inner gimbal. The inner gimbal is mounted in the outer gimbal which itself is mounted on a fixed frame as shown in Fig. When the rotor spins about X-axis with angular velocity ω rad/s and the inner gimbal precesses (rotates) about Y-axis, the spatial mechanism is forced to turn about Z-axis other than its own axis of rotation, and the gyroscopic effect is thus setup. The resistance to this motion is called gyroscopic effect.

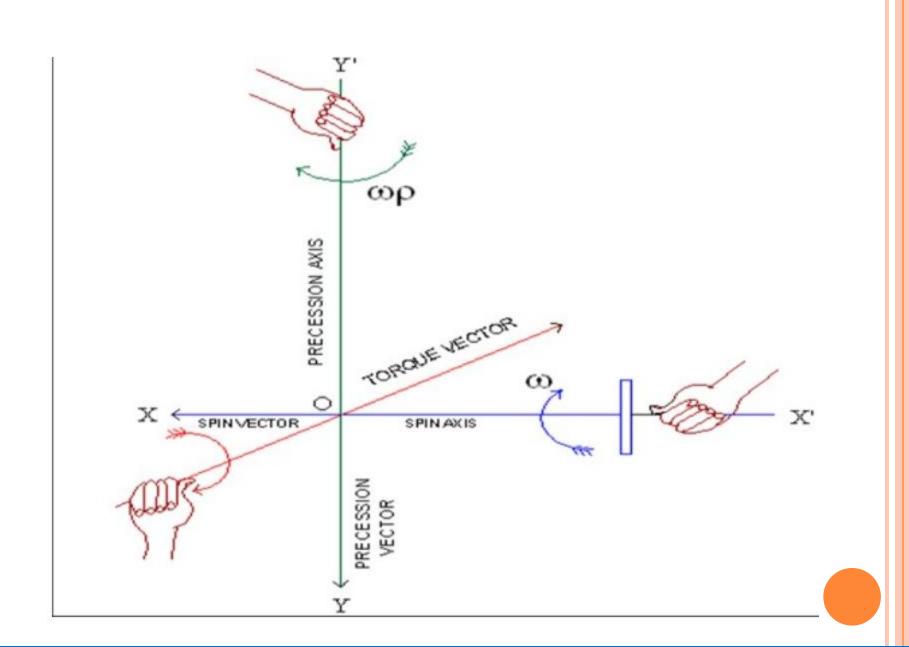


Gyroscope mechanism

GYROSCOPIC COUPLE

Consider a rotary body of mass *m having radius of gyration k mounted on the shaft* supported at two bearings. Let the rotor spins (rotates) about X-axis with constant angular velocity rad/s. The X-axis is, therefore, called spin axis, Y-axis, precession axis and Z-axis, the couple or torque axis.





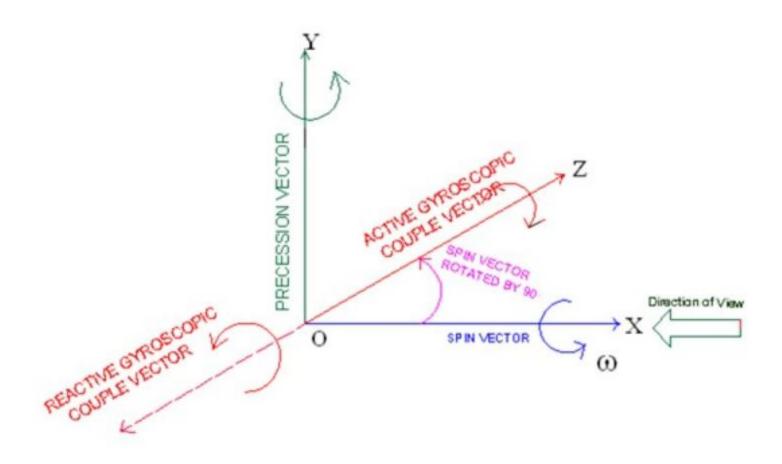


Fig. 5 Direction of active and reactive gyroscopic couple/torque vector

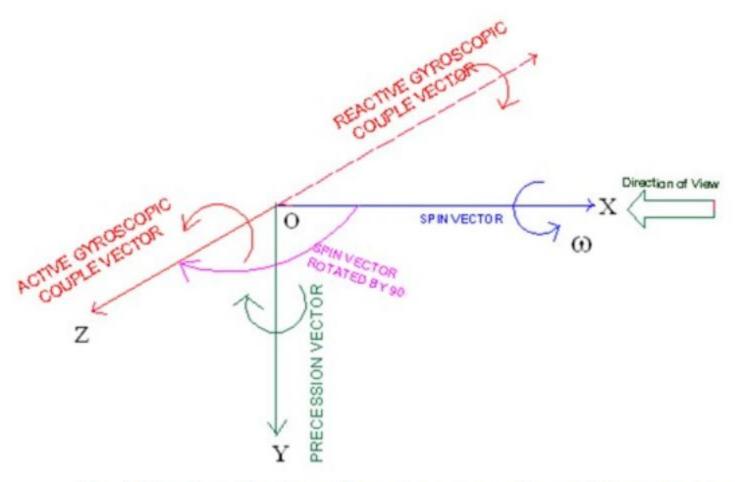


Fig. 6 Direction of active and reactive gyroscopic couple/torque vector

Problem 1

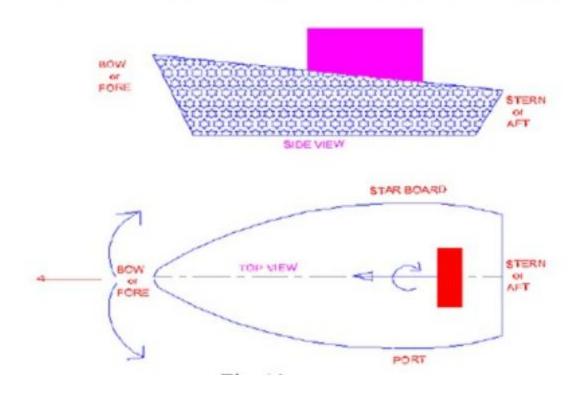
A disc of 5 kg mass with radius of gyration 70 mm is mounted at span on a horizontal shaft spins at 720 rpm in clockwise direction when viewed from the right hand bearing. If the shaft precesses about the vertical axis at 30 rpm in clockwise direction when viewed from the top, determine the reactions at each bearing due to mass of the disc and gyroscopic effect.

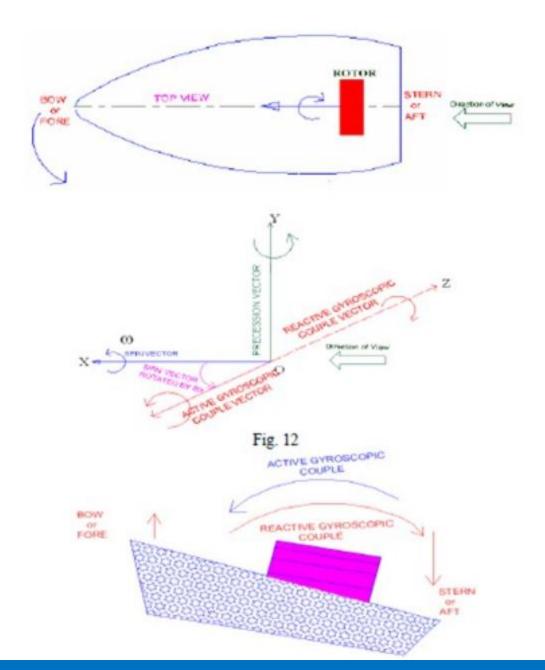
GYROSCOPIC EFFECT ON SHIP

Gyroscope is used for stabilization and directional control of a ship sailing in the rough sea. A ship, while navigating in the rough sea, may experience the following three different types of motion:

- (i) Steering—The turning of ship in a curve while moving forward
- (ii) Pitching—The movement of the ship up and down from horizontal position in a vertical plane about transverse axis
- (iii)Rolling—Sideway motion of the ship about longitudinal axis.

- (i) Bow It is the fore end of ship
- (ii) Stern It is the rear end of ship
- (iii) Starboard It is the right hand side of the ship looking in the direction of motion
- (iv) Port It is the left hand side of the ship looking in the direction of motion





Problem 2

A turbine rotor of a ship has a mass of 3500 kg and rotates at a speed of 2000 rpm. The rotor has a radius of gyration of 0.5 m and rotates in clockwise direction when viewed from the stern (rear) end. Determine the magnitude of gyroscopic couple and its direction for the following conditions

- When the ship runs at a speed of 12 knots and steers to the left in a curve of 70 m radius
- (ii) When the ship pitches 6° above and 6° below the horizontal position and the bow (Front) end is lowered. The pitching motion is simple harmonic with periodic time 30 sec.
- (iii)When the ship rolls and at a certain instant, it has an angular velocity of 0.05 rad/s clockwise when viewed from the stern

Also find the maximum angular acceleration during pitching.

Problem 3

A ship is propelled by a rotor of mass of 2000 kg rotates at a speed of 2400 rpm. The radius of gyration of rotor is 0.4 m and spins clockwise direction when viewed from bow (front) end. Find the gyroscopic couple and its effect when;

- (i) the ship takes left turn at a radius of 350 m with a speed of 35 kmph
- (ii) the ship pitches with the bow rising at an angular velocity of 1 rad/s
- (iii)the ship rolls at an angular velocity of 0.15 rad/s

Static force analysis.

If components of a machine accelerate, inertia is produced due to their masses. However, the magnitudes of these forces are small compares to the externally applied loads. Hence inertia effect due to masses are neglected. Such an analysis is known as static force analysis

• What is inertia?

 The property of matter offering resistance to any change of its state of rest or of uniform motion in a straight line is known as inertia.

conditions for a body to be in static and dynamic equilibrium?

- Necessary and sufficient conditions for static and dynamic equilibrium are
- Vector sum of all forces acting on a body is zero
- The vector sum of the moments of all forces acting about any arbitrary point or axis is zero.

Static force analysis and dynamic force analysis.

- If components of a machine accelerate, inertia forces are produced due to their masses. If the magnitude of these forces are small compared to the externally applied loads, they can be neglected while analysing the mechanism. Such an analysis is known as static force analysis.
- If the inertia effect due to the mass of the component is also considered, it is called dynamic force analysis.

D'Alembert's principle.

- D'Alembert's principle states that the inertia forces and torques, and the external forces and torques acting on a body together result in statical equilibrium.
- In other words, the vector sum of all external forces and inertia forces acting upon a system of rigid bodies is zero. The vector sum of all external moments and inertia torques acting upon a system of rigid bodies is also separately zero.

- The principle of super position states that for linear systems the individual responses to several disturbances or driving functions can be superposed on each other to obtain the total response of the system.
- The velocity and acceleration of various parts of reciprocating mechanism can be determined, both analytically and graphically.

Dynamic Analysis in Reciprocating Engines-Gas Forces

• Piston efforts (F_p): Net force applied on the piston, along the line of stroke In horizontal reciprocating engines. It is also known as effective driving force (or) net load on the gudgeon pin.

crank-pin effort.

 The component of F_Q perpendicular to the crank is known as crank-pin effort.

crank effort or turning movement on the crank shaft?

• It is the product of the crank-pin effort (F_T) and crank pin radius(r).

Forces acting on the connecting rod

- Inertia force of the reciprocating parts (F₁) acting along the line of stroke.
- The side thrust between the cross head and the guide bars acting at right angles to line of stroke.
- Weight of the connecting rod.
- \triangleright Inertia force of the connecting rod (F_C)
- \circ The radial force (F_R) parallel to crank and
- \triangleright The tangential force (F_T) acting perpendicular to crank

15.12. Determination of Equivalent Dynamical System of Two Masses by Graphical Method

Consider a body of mass m, acting at G as shown in Fig. 15.15. This mass m, may be replaced by two masses m_1 and m_2 so that the system becomes dynamical equivalent. The position of mass m, may be fixed arbitrarily at A. Now draw perpendicular CG at G, equal in length of the radius of gyration of the body, k_G . Then join AC and draw CB perpendicular to AC intersecting AG produced in B. The point B now fixes the position of the second mass m_2 .

A little consideration will show that the triangles ACG and BCG are similar. Therefore,

$$\frac{k_{\rm G}}{l_{\rm c}} = \frac{l_2}{k_{\rm G}}$$
 or $(k_{\rm G})^2 = l_1 l_2$

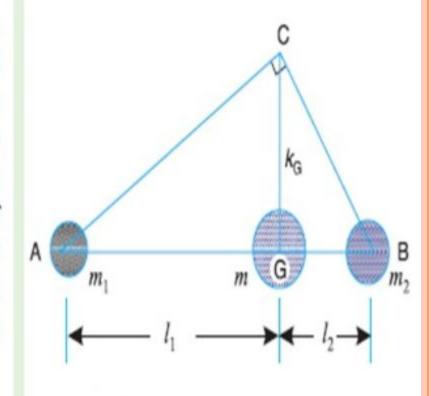


Fig. 15.15. Determination of equivalent dynamical system by graphical method.

- Determination of Equivalent Dynamical System of Two Masses by Graphical Method
- Consider a body of mass *m, acting at G as*
- o shown in fig 15.15. This mass *m, may be replaced*
- o by two masses m1 and m2 so that the system becomes dynamical equivalent. The position of mass m1 may be fixed arbitrarily at A. Now draw perpendicular CG at G, equal in length of the radius of gyration of the body, kG. Then join AC and draw CB perpendicular to AC intersecting AG produced in
- B. The point B now fixes the position of the second
- o mass *m2. T*he triangles ACG and BCG are similar. Therefo<mark>re,</mark>

Engine force Analysis:

Forces acting on the engine are weight of reciprocating masses & CR, gas forces, Friction & inertia forces (due to acceleration & retardation of engine elements)

- i) Piston effort (effective driving force)
 - Net or effective force applied on the piston.

In reciprocating engine:

The reciprocating parts (masses) accelerate during the first half of the stroke and the inertia forces tend to resist the same. Thus, the net force on the piston is reduced. During the later half of the stroke, the reciprocating masses decelerate and the inertia forces oppose this deceleration or acts in the direction of applied gas pressure and thus effective force on piston is increased.

In vertical engine, the weights of the reciprocating masses assist the piston during out stroke (down) this in creasing the piston effort by an amount equal to the weight of the piston. During the in stroke (up) piston effect is decreased by the same amount.

Force on the piston due to gas pressure; $F_P = P_1 A_1 - P_2$

 P_1 = Pressure on the cover end, P_2 = Pressure on the rod

 A_1 = area of cover end, A_2 = area of rod end, m = mass of the reciprocating parts.

Inertia force $(F_i) = m a$

$$=m.r\omega^2\left(Cos\theta + \frac{Cos2\theta}{n}\right)$$
 (Opposite to acceleration of piston)

Force on the piston $F = F_P - F_i$

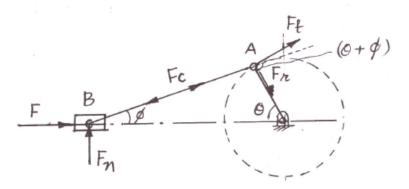
(if F_f frictional resistance is also considered)

$$F = F_P - F_i - F_i$$

In case of vertical engine, weight of the piston or reciprocating parts also acts as force.

$$\therefore$$
 F = F_P + mg - F_i - F_i

ii) Force (Thrust on the CR)



 F_c = force on the CR

Equating the horizontal components;

$$F_c \cos \phi = F \text{ or } F_c \frac{F}{\cos^2 \phi}$$

iii) Thrust on the sides of the cylinder

It is the normal reaction on the cylinder walls

$$F_n = F_c \sin \phi = F \tan \phi$$

iv) Crank effort (T)

It is the net force applied at the crank pin perpendicular to the crank which gives the required TM on the crank shaft.

$$F_{t} \times r = F_{c} r \sin(\theta + \phi)$$

$$F_{t} = F_{c} \sin(\theta + \phi)$$

$$= \frac{F}{\cos \phi} \sin(\theta + \phi)$$

v) Thrust on bearings (F_r)

The component of F_C along the crank (radial) produces thrust on bearings

$$F_r = F_c Cos(\theta + \phi) = \frac{F}{Cos \phi} Cos(\theta + \phi)$$

vi) Turning moment of Crank shaft

$$T = F_t \times r$$

$$= \frac{F}{\cos \phi} \sin(\theta + \phi) \times r = \frac{F_r}{\cos \phi} \left(\sin \theta + \cos \phi + \cos \theta \sin \phi \right)$$

$$= F \times r \left(\sin \theta + \cos \theta \frac{\sin \phi}{\cos \phi} \right)$$
Proved earlier
$$= F \times r \left(\sin \theta + \cos \theta \frac{\sin \theta}{\cos \phi} \right)$$

$$= F \times r \left(\sin \theta + \cos \theta \frac{\sin \theta}{\cos \phi} \right)$$

$$= F \times r \left(\sin \theta + \cos \theta \frac{\sin \theta}{\cos \phi} \right)$$
Proved earlier

$$\sin \phi = \frac{\sin \theta}{n}$$

$$= F \times r \left(\sin \theta + \frac{\sin 2 \theta}{2\sqrt{n^2 - \sin^2 \theta}} \right)$$

Also,

$$r\sin(\theta + \phi) = OD\cos\phi$$

$$T = F_{t} \times r$$

$$= \frac{F}{\cos \phi}. r \sin (\theta + \phi)$$

$$= \frac{F}{\cos \phi}. \ OD \ \cos \phi$$

$$T = F \times OD$$
.

TURNING MOMENT DIAGRAMS AND FLYWHEEL

• A flywheel is nothing but a rotating mass which is used as an energy reservoir in a machine which absorbs the energy when the speed in more and releases the energy when the speed is less, thus maintaining the fluctuation of speed within prescribed limits.

Difference between Governor and Flywheel:

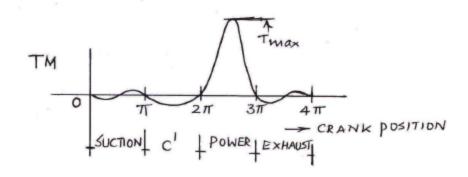
A governor controls the speed of the output shaft within close limits, but its action depends upon controlling the amount of working fluid to the engine as required by the load on the engine. The flywheel, on the other hand, serves only to smooth out the energy transfer in each energy cycle. For example, if an engine is operating at quarter load, with the governor in a particular position controlling the amount of working media to the engine; the flywheel would take care of redistributing the energy throughout a cycle. If the load was increased to full load the governor action would permit more working fluid to the engine maintaining the speed of the engine, but when balance of working fluid to the engine and load on the engine was reached, the flywheel would continue its action of redistributing the energy throughout a cycle. Changes of seed in an engine will cause the governor to respond and attempt to do the flywheels job. Usually, the effect of the governor is disregarded in the design of the flywheel. The flywheel analysis is limited to engines receiving power at a variable rate and delivering it to a shaft at an approximately constant rate.



Uses of turning moment Diagram

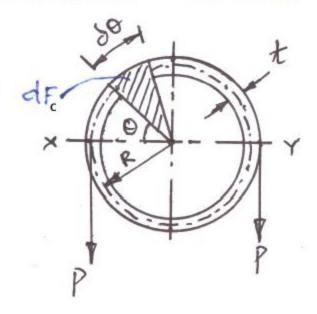
- The area under the turning moment diagram represents work done per cycle. The area multiplied by number of cycles per second gives the power developed by the engine.
- By dividing the area of the turning moment diagram with the length of the base we get the mean turning moment. This enables us the find the fluctuation of energy.
- The max. ordinate of the turning moment diagram gives the maximum torque to which the crank shaft is subjected. This enables us the find diameter of the crank shaft.

TMD for a four stroke I.C. Engine



Size of fly wheel and hoop stress developed in a fly wheel.

Consider a rim of the fly wheel as shown in figure. Let D = mean diameter of rim, R = mean radius of rim, t = thickness of the fly wheel, A = cross sectional as area of rim in m² and ρ be the density of the rim material in Kg/m³, N be the speed of the fly wheel in rpm, ω = angular velocity in rad/sec, V = linear velocity in m/ σ , hoop stress in N/m² due to centrifugal force.



Consider small element of the rim. Let it subtend an angle $\delta\theta$ at the centre of flywheel.

Volume of the small element = $R\delta\theta$.A.

Mass of the small element = $dm = R\delta\theta$. A ρ

The centrifugal force on the small element

$$dF_C = dm\omega^2 R$$

= $R\delta\theta . A\omega^2 R \rho$
= $R^2 A.\omega^2 \delta\theta \rho$

Resolving the centrifugal force vertically

$$dF_C = dF_C Sin\theta$$

$$= \rho R^2 A \omega^2 Sin\theta. \delta\theta \quad --- (1)$$

Total Vertical upward force across diameter X & Y

$$= \int_0^{\infty} \rho R^2 A \omega^2 Sin\theta. \, \delta\theta$$
$$= \rho R^2 A \omega^2 \int_0^{\infty} Sin\theta. \, \delta\theta$$
$$2\rho = 2\rho A R^2 \omega^2$$

This vertical upward force will produce tensile stress on loop stress developed & it is resisted by 2P.

We know that,
$$\sigma = P/A$$

$$P = \sigma A$$

$$\therefore 2P = 2\sigma A$$

$$PAR^{2}\omega^{2} = 2\sigma A$$

$$\sigma = \rho R^{2}\omega^{2}$$
 % up to this deviation

Also,

Linear velocity V=Rx
$$\omega$$

$$\sigma = \delta V^2$$

$$V = \sqrt{\sigma/\delta}$$

Mass of the rim = volume x density

$$m = \pi dA \times \rho$$

Problem 1:

A shaft fitted with a flywheel rotates at 250 r.p.m. and drives a machine. The torque of the machine varies in a cyclic manner over a period of 3 revolutions. The torque rises from 750 N-m to 3000 N-m uniformly during ½ revolution and remains constant fore the following revolution. It then falls uniformly to 750 N-m during the next ½ revolution and remains constant for one revolution, the cycle being repeated thereafter.

Determine the power required to drive the machine and percentage fluctuation in speed, if the driving torque applied to the shaft is constant and the mass of the flywheel is 500 kg with radius of gyration of 600 mm.

Solution.

Given:
$$N = 250 \text{ r.p.m}$$
 or $\omega = 2\pi \times 250/60 = 26.2 \text{ rad/s}$; $m = 500 \text{kg}$; $k = 600 \text{ m} = 0.6$

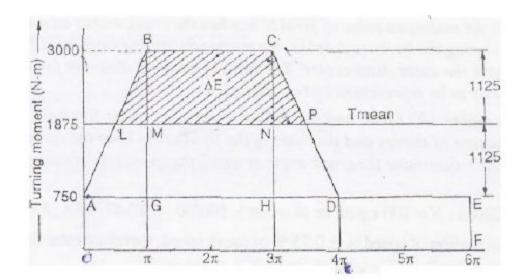
The turning moment diagram for the complete cycle is drawn.

The torque required for one complete cycle

= Area of figure OABCDEF
= Area OAEF + Area ABG + AreaBCHG + Area CDH
=
$$OF \times OA + \frac{1}{2} \times AG \times BG + GH \times CH + \frac{1}{2} \times HD \times CH$$

= $6\pi \times 750 + \frac{1}{2} \times \pi (3000 - 750) + 2\pi (3000 - 750) + \frac{1}{2} \times \pi (3000 - 750)$
= $11250 \pi N - m$

Torque required for one complete cycle = $T_{mean} \times \pi N - m$



Power required to drive the machine, $P = T_{mean} \times \omega = 11875 \times 26.2 = 49125W = 49.125kW$.

To find Coefficient of fluctuation of speed, δ .

Find the values of LM and NP.

From similar triangles ABG and BLM,

$$\frac{LM}{AG} = \frac{BM}{BG}$$
 or $\frac{LM}{\pi} = \frac{3000 - 1875}{3000 - 750} = 0.5$ or $LM = 0.5\pi$

From similar triangles CHD and CNP,

$$\frac{NP}{HD} = \frac{CN}{CH} \text{ or } \frac{NP}{\pi} = \frac{3000 - 1875}{3000 - 750} = 0.5 \text{ or } NP = 0.5\pi$$

From the figure, we find that,

The area above the mean torque line represents the maximum fluctuation of energy. Therefore the maximum fluctuation of energy, ΔE

$$=$$
 Area $LBCP =$ Area $LBM +$ Area $MBCN +$ Area PNC

$$= \frac{1}{2} \times LM \times BM + MN \times BM + \frac{1}{2} \times NP \times CN$$

$$= \frac{1}{2} \times 0.5 \ \pi \times 1125 + 2\pi \times 1125 + \frac{1}{2} \times 0.5 \ \pi \times 1125 = 8837 \ N - m$$

We know that maximum fluctuation of energy (ΔE),

$$8837 = m.k^2.\omega^2.\delta = 500 (0.6)^2 (26.2)^2 \delta = 123 559 \delta$$

$$\delta = 0.071$$

Problem 2

The torque delivered by two stroke engine is represented by $T = 1000+300 \sin 2\theta-500 \cos 2\theta$ where θ is angle turned by the crack from inner dead under the engine speed. Determine work done per cycle and the power developed.

Solution

θ , deg.	T, N-m
0	500
90	1500
180	500
270	1500
360	500

Work done / cycle = Area under the turning moment diagram.

$$= \int_{0}^{2\pi} T \ d\theta$$

$$= \int_{0}^{2\pi} (1000 + 300\sin 2\theta - 500\cos 2\theta) \ d\theta$$

$$= 2000\pi \ N - m$$

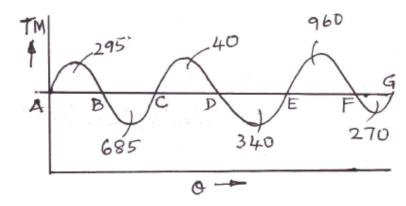
$$T_{mean} = \frac{W.D / cycle}{2\pi}$$

$$= \frac{2000\pi}{2\pi} = 1000 \ N - m$$

Power developed =
$$T_{mean} \times \omega_{mean}$$

$$= 1000 \times \frac{2\pi N}{60}$$
$$= 1000 \times \frac{2\pi \times 200}{60}$$
$$= 26179W$$

The TMD for a petrol engine is drawn to the following scale, turning moment, 1mm = 5Nm, crank 1mm = 1°. The TMD repeats itself at every half revolution of the engine & areas above & below the mean turning moment line taken in order are 295, 685, 40, 340, 960, 270 mm². The rotating parts are equivalent to a mass of 36 kg at a radius of gyration of 150mm. Calculate the maximum fluctuation of energy & co-efficient of fluctuation of speed when engine runs at 1800rpm



Energy at
$$A = E$$

Energy at $B = E + a_1$
 $= E + 295$
Energy at $C = E + 295 - 685 = E - 390$
Energy at $D = E + 295 - 685 + 40 = E - 350$
Energy at $E = E - 350 - 340 = E - 690$
Energy at $F = E - 690 + 960 = E + 270$
Energy at $G = E + 270 - 270 = E$
 $\therefore A = G$

Max Energy =
$$E + 295$$

Min Energy = $E - 690$

$$m = 36kg, k = 150mm, N = 1800rpm$$

Maximum Fluctuation of Energy $\Delta E = E + 295 - (E - 690)$

$$=985mm^{2}$$

Scale: $1mm = 5Nm \& 1mm = 1^{\circ}$

$$Torque \times \theta = \frac{5}{180}\pi \times 1 = \frac{\pi}{36}Nm$$

$$\Delta E = mk^2 \omega^2 \delta$$

$$86 = 36 \times 0.15^2 \times \left(\frac{2\Pi(1800)}{60}\right)^2 \delta$$

 $\delta = 0.003$ or 0.3%

The TMD for a multi cylinder engine has been drawn to a scale 1mm to 500Nm torque & 1mm to 6° of crank displacement. The intercepted area in order from one end is mm² are -30, 410, -280, 320, -330, 250, -360, +280, -260 mm² when engine is running at 800rpm. The engine has a stroke of 300mm & fluctuation of speed is not to exceed $\pm 2\%$ of the mean speed, determine

 a suitable diameter & cross section of the fly wheel rim for a limiting value of the safe centrifugal stress of 7MPa. The material density may be assumed as 7200 kg/m³. The width of the rim is to be 5times the thickness.

Solution:

$$N = 800 \ rpm$$

 $\pm 2 \ \% means, \ \delta = 4\% = 0.04 \ T$
 $\sigma = 7 \ Mpa = 7 \ N / m2$
 $\rho = 7 \ 200 \ kg / m^3$
 $0 = 320 \ 280 \ 330 \ 360 \ 260$

Energy at A = EEnergy at B = E - 30Energy at C = E - 30 + 410 = E + 380Energy at D = E + 380 - 280 = E + 100Energy at E = E + 100 + 320 = E + 420Energy at F = E + 420 - 330 = E + 90Energy at G = E + 90 + 250 = E + 340Energy at H = E + 340 - 360 = E - 20Energy at I = E - 20 + 280 = E + 260Energy at J = E + 260 - 260 = E $\Delta E = E + 420 - (E - 30)$ $=450mm^{2}$ 1mm = 500Nm, $1mm = 6^{\circ} (0.1047 \ radians)$, $1mm^2 = 52.35Nm$ $\Delta E = 450 \times 52.35 = 23557.5 \text{ Nm}$ $\sigma = \rho V^2$ $\Delta E = mr^2 \omega^2 \delta$ $V = \frac{\pi DN}{60}$, D = 0.745 m $7 \times 10^6 = 7200 \ V^2 = mV^2 \delta$ $V = r\omega$ Cross sectional area A = bt $V = 31.18 \ m/s$ $A = (5t)t = 5t^2$ Fluctuation of energy $\Delta E = mV^2 \delta$ $23.56 \times 10^3 = m(31.18)^2 (0.04)$ $m = 605 \, kg$ $m = Volume \times Density$ $\pi DA \times \rho$ $605 = \pi(0.745)(5t^2)7200$ t = 0.084 m

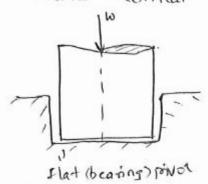
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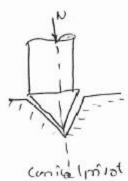
Area = $5t^2 = 0.035m^2$

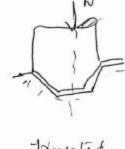
PIVOT BEARING

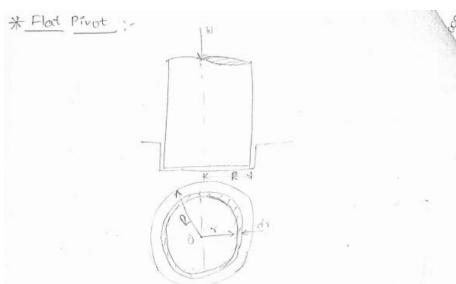
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The votating chastis are diequently subjected to avial thrust. These chastis can be kept in correct anial position if bearing surfaces are provided. The bearing surfaces which are that (on) conical carry the original thrust. The bearing surfaces placed at the end of a shaft are known as pivoli. The pivol may that, conical the truncated conical surfaces.









The bearing surjace placed at the end of shaft is known as privot. If the surjace is that ais shown, then bearing surjace is called that mid 60-bot - step. There will be thetim along the surjace of contact between enall a bearing. The power lost can be obtained by calculating largue.

let, 14 = thial load, (a) load transmitted to the bearing surface

R > ladius of pivot:

p = co-effecient of friction.

p > Inlensity of pr = NIm.

Total trictnal largue.

Y > Ladius of ring.

```
consider a circular ring of theres y & thickness
           shown.
       as
            .. Alea of ring = anrida.
  He will consider 2 cases; namely;
     in Uniform Pressure over bearing sugare of
      (ii) Uniform Near over bearing suface
i) case of Uniform Pr. !
   When the Pa is assumed to be uniform over
The bearing surface, then intentily of pressure is
                  P= Asial load N - (1)
Accorded TR2
     given by.
Now, the load Francisted to the ring f-frictimal
  largue on the ring,
         wad Franchitted to the ring,
                 dN= Pr ming x According
                    > px 2 Ardr.
  frictional -force on ring,
                     df = NxdW
                        . pex load in ring
(Frictional loique on the ring) Howert of frictnet force ingrazis.
                        · hx bx218dr.
                     dT = frictional -love x Radius of ring
                         . dfxx.
             .. di . Mx px 2 h. r. d r. r
                     · 4. px 27, dr - - @
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Now, the total factoral 16 gove 1011 be obtained by integrating above ex. \mathbb{Q} .

Total factional 16 gove, $T = \begin{pmatrix} R_{2}\pi \mu p^{3} dr \\ 2\pi \mu p \end{pmatrix}^{R} r^{2} dr$ $= 2\pi \mu p \begin{pmatrix} r^{3} \\ 3 \end{pmatrix}^{R}$

and Transmitted to the ring, · Dr + Alea of ring , px 271.dx C c 27 r.dr dw : 27c.dy - 6. Total load transmitted to the bearing, is obtained by interesting from 016 R Total load transmitting to the bearing, M= JRdIN . 1 2 2 Cdr = 2 TC 1 dr = 2 TC [1] 0 W. 27CR C= W 27R Non frictimal torque in the ring, of F= Hx load mring = pxdw LIX 27 Cdr

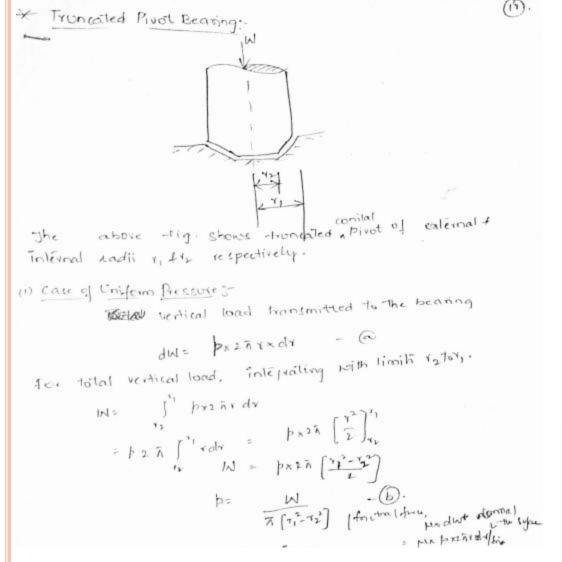
Problems: find the power lost in thickin assuming of 1) Uniterm pr. I (ii) Uniterm wear. when a vertical shapl of 100mm dia. rotating at 1501pm testion a flat end-foot step bearing. The coeffecient of friction is equal to 0.05 f shaft carries a vertical load of 15 km.

Soli- Given.

Dia, D= 100 mm=) 0.1m :. R=0.1_ 0.05m N=150 ipm; Co-effectent of fiction, N=0.05 10ad, N=15KN = 15K10²N.

is Power lost in fiction assuming uniform pressure.

For uniform Pr.



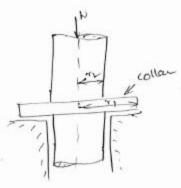
$$T: \int_{1}^{1} px px 2 \overline{x} x x \frac{dx}{dx} x$$

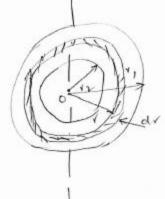
$$= \frac{2 \overline{x} x \mu \cdot p}{8 \overline{x} x^{3}} \int_{1}^{1} x^{2} \frac{dx}{8 \overline{x} x^{3}} \int_{1}^{1} x^{3} \frac{dx}{$$

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=) T = 1 3/1. H. W [(1,2-1)]

Passifien with shaft (but not at the end) Licarry anial though is known as collar. Collar bearings are also known as -throst bearings





1ct, The External radius of collar

12 - Internal radius of collar

12 - Internal radius of collar

13 - Internal radius of collar

14 - Axial coad a total load transmitted to beamy enfau

15 - co-effection of friction

15 - Total frictional largue.

16 - Total frictional largue.

16 - Consider a circular sing of radius in thickness dr

16 - Acea of ring, = 2 no. dr.

16 - Load m ring, = pr x Acea of ring

16 - Load m ring, = pr x Acea of ring

friction lorgue = frictindorce x Radius

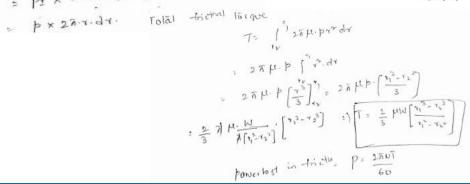
p. p. x 2 x rd x x r

= 2 x pp r. dr.

- total trictimal lorgue,

T = 1 2 x p. pr. dr.

(i) Uniform Piessore: p = constantTotal load It ansmitted to the bearing. $W = \int_{1}^{1} load m may (dM)$ $= \int_{1}^{1}$



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Parts constant

STATIC AND DYNAMIC BALANCING

When man invented the wheel, he very quickly learnt that if it wasn't completely round and if it didn't rotate evenly about it's central axis, then he had a problem!

What the problem he had?

The wheel would vibrate causing damage to itself and it's support mechanism and in severe cases, is unusable.

A method had to be found to minimize the problem. The mass had to be evenly distributed about the rotating centerline so that the resultant vibration was at a minimum.

UNBALANCE:

The condition which exists in a rotor when vibratory force or motion is imparted to its bearings as a result of centrifugal forces is called unbalance or the uneven distribution of mass about a rotor's rotating centreline.

BALANCING:

Balancing is the technique of correcting or eliminating unwanted inertia forces or moments in rotating or reciprocating masses and is achieved by changing the location of the mass centres.

The objectives of balancing an engine are to ensure:

- That the centre of gravity of the system remains stationery during a complete revolution of the crank shaft and
- That the couples involved in acceleration of the different moving parts balance each other.

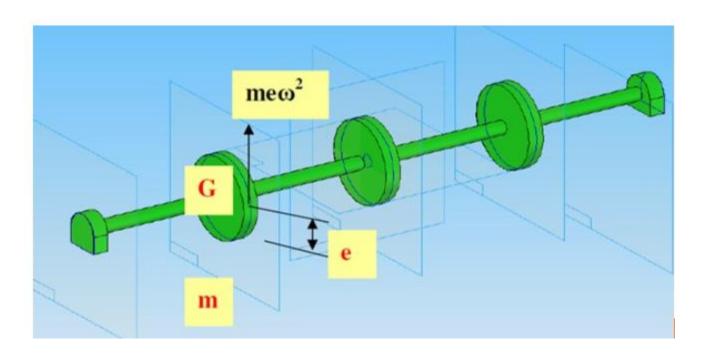
Types of balancing:

- a) Static Balancing:
- Static balancing is a balance of forces due to action of gravity.
- ii) A body is said to be in static balance when its centre of gravity is in the axis of rotation.
- b) Dynamic balancing:
- i) Dynamic balance is a balance due to the action of inertia forces.
- ii) A body is said to be in dynamic balance when the resultant moments or couples, which involved in the acceleration of different moving parts is equal to zero.
- iii) The conditions of dynamic balance are met, the conditions of static balance are also met.

BALANCING OF ROTATING MASSES

When a mass moves along a circular path, it experiences a centripetal acceleration and a force is required to produce it. An equal and opposite force called centrifugal force acts radially outwards and is a disturbing force on the axis of rotation. The magnitude of this remains constant but the direction changes with the rotation of the mass.

In a revolving rotor, the centrifugal force remains balanced as long as the centre of the mass of rotor lies on the axis of rotation of the shaft. When this does not happen, there is an eccentricity and an unbalance force is produced. This type of unbalance is common in steam turbine rotors, engine crankshafts, rotors of compressors, centrifugal pumps etc.



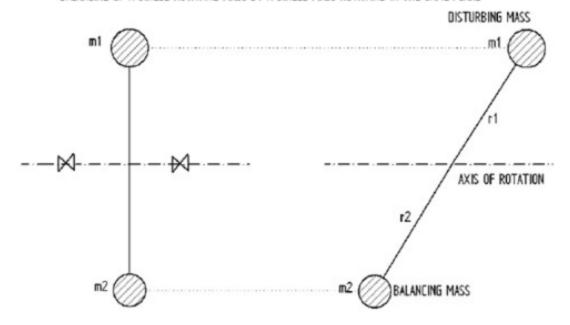
The unbalance forces exerted on machine members are time varying, impart vibratory motion and noise, there are human discomfort, performance of the machine deteriorate and detrimental effect on the structural integrity of the machine foundation.

Balancing involves redistributing the mass which may be carried out by addition or removal of mass from various machine members. Balancing of rotating masses can be of

- Balancing of a single rotating mass by a single mass rotating in the same plane.
- Balancing of a single rotating mass by two masses rotating in different planes.
- 3. Balancing of several masses rotating in the same plane
- 4. Balancing of several masses rotating in different planes

BALANCING OF A SINGLE ROTATING MASS BY A SINGLE MASS ROTATING IN THE SAME PLANE

BALANCING OF A SINGLE ROTATING MASS BY A SINGLE MASS ROTATING IN THE SAME PLANE



Consider a disturbing mass m1 which is attached to a shaft rotating at rad/s.

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r = radius of rotation of the mass m

The centrifugal force exerted by mass m1 on the shaft is given by, F = m r c 1 1

This force acts radially outwards and produces bending moment on the shaft. In order to counteract the effect of this force Fc1, a balancing mass m2 may be attached in the same plane of rotation of the disturbing mass m1 such that the centrifugal forces due to the two masses are equal and opposite.

BALANCING OF A SINGLE ROTATING MASS BY TWO MASSES ROTATING

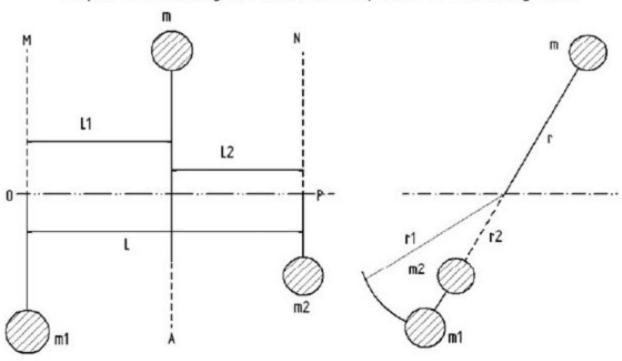
There are two possibilities while attaching two balancing masses:

- The plane of the disturbing mass may be in between the planes of the two balancing masses.
- The plane of the disturbing mass may be on the left or right side of two planes containing the balancing masses.

In order to balance a single rotating mass by two masses rotating in different planes which are parallel to the plane of rotation of the disturbing mass i) the net dynamic force acting on the shaft must be equal to zero, i.e. the centre of the masses of the system must lie on the axis of rotation and this is the condition for static balancing ii) the net couple due to the dynamic forces acting on the shaft must be equal to zero, i.e. the algebraic sum of the moments about any point in the plane must be zero. The conditions i) and ii) together give dynamic balancing.

THE PLANE OF THE DISTURBING MASS LIES IN BETWEEN THE PLANES OF THE TWO BALANCING MASSES.

The plane of the disturbing mass lies inbetween the planes of the two balancing masses



Consider the disturbing mass m lying in a plane A which is to be balanced by t rotating masses m₁ and m₂ lying in two different planes M and N which are parallel the plane A as shown.

Let r, r₁ and r₂ be the radii of rotation of the masses in planes A, M and N respectively. Let L₁, L₂ and L be the distance between A and M, A and N, and M and N respectively Now,

The centrifugal force exerted by the mass m in plane A will be,

Similarly,

The centrifugal force exerted by the mass m1 in plane M will be,

And the centrifugal force exerted by the mass m2 in plane N will be,

For the condition of static balancing,

Now, to determine the magnitude of balancing force in the plane 'M' or the dynamic force at the bearing 'O' of a shaft, take moments about 'P' which is the point of intersection of the plane N and the axis of rotation.

Therefore,

$$F_{c1} \times L = F_c \times L_2$$
or $m_1 \omega^2 r_1 \times L = m \omega^2 r \times L_2$

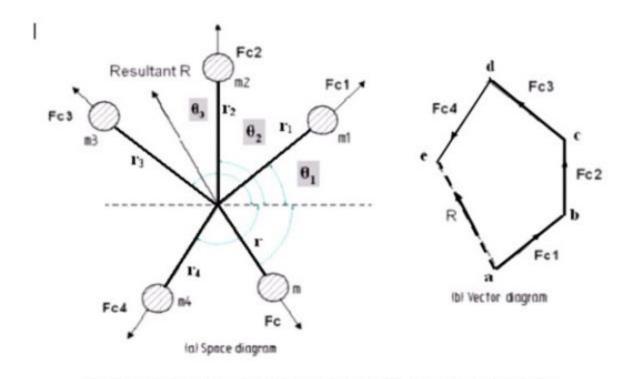
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Similarly, in order to find the balancing force in plane 'N' or the dynamic force at the bearing 'P' of a shaft, take moments about 'O' which is the point of intersection of the plane M and the axis of rotation.

Therefore,

For dynamic balancing equations (5) or (6) must be satisfied along with equation (4).

Balancing Multi-cylinder Engines, Balancing V-engines



BALANCING OF SEVERAL MASSES ROTATING IN THE SAME PLANE

Consider a rigid rotor revolving with a constant angular velocity ω rad/s. A number of masses say, four are depicted by point masses at different radii in the same transverse plane.

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If m₁, m₂, m₃ and m₄ are the masses revolving at radii r₁, r₂, r₃ and r₄ respectively in the same plane.

The centrifugal forces exerted by each of the masses are F_{c1}, F_{c2}, F_{c3} and F_{c4} respectively. Let F be the vector sum of these forces, i.e.

The rotor is said to be statically balanced if the vector sum F is zero. If the vector sum F is not zero, i.e. the rotor is unbalanced, then introduce a counterweight (balance weight) of mass 'm' at radius 'r' to balance the rotor so that,

$$m_{_{1}} \omega^{2} r_{_{1}} + m_{_{2}} \omega^{2} r_{_{2}} + m_{_{3}} \omega^{2} r_{_{3}} + m_{_{4}} \omega^{2} r_{_{4}} + m \omega^{2} r = 0 - - - - - - - - (2)$$
or
$$m_{_{1}} r_{_{1}} + m_{_{2}} r_{_{2}} + m_{_{3}} r_{_{3}} + m_{_{4}} r_{_{4}} + m r = 0 - - - - - - - - - - - - (3)$$

The magnitude of either 'm' or 'r' may be selected and the other can be calculated. In general, if $\sum \mathbf{m_i} \mathbf{r_i}$ is the vector sum of $\mathbf{m_1} \mathbf{r_1}$, $\mathbf{m_2} \mathbf{r_2}$, $\mathbf{m_3} \mathbf{r_3}$, $\mathbf{m_4} \mathbf{r_4}$ etc, then,

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1. Analytical Method:

Procedure:

Step 1: Find out the centrifugal force or the product of mass and its radius of rotation exerted by each of masses on the rotating shaft, since ω^2 is same for each mass, therefore the magnitude of the centrifugal force for each mass is proportional to the product of the respective mass and its radius of rotation.

Step 2: Resolve these forces into their horizontal and vertical components and find their sums, i.e.,

Sum of the horizontal components

Sumof the vertical components

Step 3: Determine the magnitude of the resultant centrifugal force

$$R = \sqrt{\left(\sum_{i=1}^{n} m_{i} r_{i} \cos \theta_{i}\right)^{2} + \left(\sum_{i=1}^{n} m_{i} r_{i} \sin \theta_{i}\right)^{2}}$$

Step 4: If θ is the angle, which resultant force makes with the horizontal, then

$$tan\theta = \frac{\sum_{i=1}^{n} m_{i} r_{i} \sin \theta_{i}}{\sum_{i=1}^{n} m_{i} r_{i} \cos \theta_{i}}$$

Step 5: The balancing force is then equal to the resultant force, but in opposite direction.

Step 6: Now find out the magnitude of the balancing mass, such that

Where, m = balancing mass and r = its radius of rotation

2. Graphical Method:

Step 1:

Draw the space diagram with the positions of the several masses, as shown.

Step 2:

Find out the centrifugal forces or product of the mass and radius of rotation exerted by each mass.

Step 3:

Now draw the vector diagram with the obtained centrifugal forces or product of the masses and radii of rotation. To draw vector diagram take a suitable scale. Let ab, bc, cd, de represents the forces F_{c1} , F_{c2} , F_{c3} and F_{c4} on the vector diagram. Draw 'ab' parallel to force F_{c1} of the space diagram, at 'b' draw a line parallel to force

Step 4:

As per polygon law of forces, the closing side 'ae' represents the resultant force in magnitude and direction as shown in vector diagram.

Step 5:

The balancing force is then, equal and opposite to the resultant force.

 F_{c2} . Similarly draw lines cd, de parallel to F_{c3} and F_{c4} respectively.

Step 6:

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Determine the magnitude of the balancing mass (m) at a given radius of rotation (r), such that,

$$F_c = m\omega^2 r$$
 or
$$or$$

$$mr = resultant of m_1 r_1, m_2 r_2, m_3 r_3 and m_4 r_4$$

GOVERNORS

GOVERNORS

Engine Speed control

This presentation is from Virginia Tech and has not been edited by Georgia Curriculum Office.

GOVERNORS

- Governors serve three basic purposes:
- Maintain a speed selected by the operator which is within the range of the governor.
- Prevent over-speed which may cause engine damage.
- Limit both high and low speeds.

GOVERNORS

- Generally governors are used to maintain a fixed speed not readily adjustable by the operator or to maintain a speed selected by means of a throttle control lever.
- In either case, the governor protects against overspeeding.

HOW DOES IT WORK?

- If the load is removed on an operating engine, the governor immediately closes the throttle.
- If the engine load is increased, the throttle will be opened to prevent engine speed form being reduced.

EXAMPLE

o The governor on your lawnmower maintains the selected engine speed even when you mow through a clump of high grass or when you mow over no grass at all.



HUNTING

- Hunting is a condition whereby the engine speed fluctuate or is erratic usually when first started.
- The engine speeds up and slows down over and over as the governor tries to regulate the engine speed.
- This is usually caused by an improperly adjusted carburetor.

STABILITY

- Stability is the ability to maintain a desired engine speed without fluctuating.
- Instability results in hunting or oscillating due to over correction.
- Excessive stability results in a dead-beat governor or one that does not correct sufficiently for load changes.

SENSITIVITY

- Sensitivity is the percent of speed change required to produce a corrective movement of the fuel control mechanism.
- High governor sensitivity will help keep the engine operating at a constant speed.

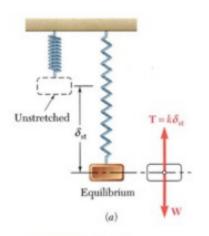
SUMMARY

- Small engine governors are used to:
 - Maintain selected engine speed.
 - Prevent over-speeding.
 - Limit high and low speeds.

SUMMARY

 The governor must have stability and sensitivity in order to regulate speeds properly. This will prevent hunting or erratic engine speed changes depending upon load changes. Mechanical vibration is the motion of a particle or body which oscillates about a position of equilibrium. Most vibrations in machines and structures are undesirable due to increased stresses and energy losses.

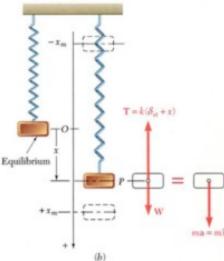
FREE VIBRATIONS OF PARTICLES. SIMPLE HARMONIC MOTION



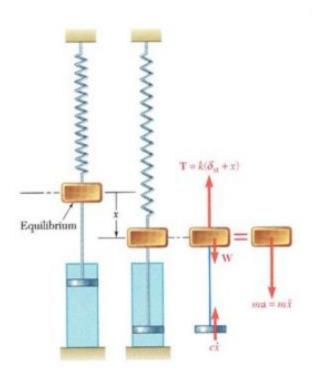
 If a particle is displaced through a distance x_m from its equilibrium position and released with no velocity, the particle will undergo simple harmonic motion,

$$ma = F = W - k(\delta_{st} + x) = -kx$$

$$m\ddot{x} + kx = 0$$



DAMPED FREE VIBRATIONS



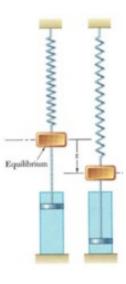
 All vibrations are damped to some degree by forces due to dry friction, fluid friction, or internal friction.

DAMPED FREE VIBRATIONS

· Characteristic equation,

$$m\lambda^2 + c\lambda + k = 0$$
 $\lambda = -\frac{c}{2m} \pm \sqrt{\left(\frac{c}{2m}\right)^2 - \frac{k}{m}}$ $c_c = 2m\omega_n = \text{critical damping coefficient}$

DAMPED FORCED VIBRATIONS



$$m\ddot{x} + c\dot{x} + kx = P_m \sin \omega_f t$$

$$m\ddot{x} + c\dot{x} + kx = P_m \sin \omega_f t \qquad \qquad x = x_{complementary} + x_{particular}$$